## TABLE OF CONTENTS

## **SECTION 3**

## EMERGENCY PROCEDURES

Paragraph	
	No.
General	3-1
	3-2
	3-2
Engine Fire During Start	3-2
Engine Power Loss During Takeoff	3-2
Engine Power Loss In Flight	3-3
Power Off Landing	3-3
Fire In Flight	3-4
Loss of Oil Pressure	3-4
Loss of Fuel Pressure	3-4
High Oil Temperature	3-5
Electrical Failures	3-5
Electrical Overload	3-6
Spin Recovery	3-6
Open Door	3-7
Carburetor Icing	3-7
Engine Roughness	3-7
Amplified Emergency Procedures (General)	3-9
Engine Fire During Start	3-9
Engine Power Loss During Takeoff	3-9
Engine Power Loss In Flight	3-10
Power Off Landing	3-11
Fire In Flight	3-11
Loss of Oil Pressure	3-12
Loss of Fuel Pressure	3-13
High Oil Temperature	3-13
	General Airspeeds for Safe Operation Emergency Procedures Checklist Engine Fire During Start Engine Power Loss During Takeoff Engine Power Loss In Flight Power Off Landing Fire In Flight Loss of Oil Pressure Loss of Fuel Pressure High Oil Temperature Electrical Failures Electrical Overload Spin Recovery Open Door Carburetor Icing Engine Roughness Amplified Emergency Procedures (General) Engine Power Loss In Flight Power Off Landing Fire In Flight Loss of Oil Pressure Loss of Oil Pressure Loss of Oil Pressure Loss of Oil Pressure

## TABLE OF CONTENTS (cont)

## SECTION 3 (cont)

Parag No.	raph	Page No.
3.25	Electrical Failures.	3-14
3.27	Electrical Overload	3-14
3.29	Spin Recovery	3-15
3.31	Open Door	3-15
3.33	Carburetor Icing	3-16
3.35	Engine Roughness	3-16

#### SECTION 3

#### EMERGENCY PROCEDURES

#### 3.1 GENERAL

The recommended procedures for coping with various types of emergencies and critical situations are provided by this section. All of required (FAA regulations) emergency procedures and those necessary for the operation of the airplane as determined by the operating and design features of the airplane are presented.

Emergency procedures associated with those optional systems and equipment which require handbook supplements are provided in Section 9 (Supplements).

The first portion of this section consists of an abbreviated emergency check list which supplies an action sequence for critical situations with little emphasis on the operation of systems.

The remainder of the section is devoted to amplified emergency procedures containing additional information to provide the pilot with a more complete understanding of the procedures.

These procedures are suggested as a course of action for coping with the particular condition described, but are not a substitute for sound judgment and common sense. Pilots should familiarize themselves with the procedures given in this section and be prepared to take appropriate action should an emergency arise.

Most basic emergency procedures, such as power off landings, are a normal part of pilot training. Although these emergencies are discussed here, this information is not intended to replace such training, but only to provide a source of reference and review, and to provide information on procedures which are not the same for all aircraft. It is suggested that the pilot review standard emergency procedures periodically to remain proficient in them.

ISSUED: JULY 12, 1995

3.3 AIRSPEEDS FOR SAFE OPERATION
Stall Speeds 2550 lbs (0° Flaps)
2550 lbs (Full Flaps)
2550 lbs
Never Exceed Speed
2550 lbs (0° Flaps)
3.5 EMERGENCY PROCEDURES CHECK LIST
ENGINE FIRE DURING START
Starter
Electric fuel pump
Abandon if fire continues.
ENGINE POWER LOSS DURING TAKEOFF
If sufficient runway remains for a normal landing, land straight ahead.
If insufficient runway remains:  Maintain safe airspeed.  Make only shallow turn to avoid obstructions.
Flaps as situation requires.
If sufficient altitude has been gained to attempt a restart:  Maintain safe airspeed.
Fuel selector
Electric fuel pump check ON Mixture check RICH Carburetor heat ON
If power is not regained, proceed with power off landing.

ENGINE POWER LOSS IN FLIGHT
If at low altitude:  Airspeed
If altitude permits:  Fuel selector
Electric fuel pump ON Mixture RICH Carburetor heat ON Engine gauges check for indication of cause of power loss
If no fuel pressure is indicated, check tank selector position to be sure it is on a tank containing fuel.
When power is restored:  Carburetor heat
POWER OFF LANDING
Trim for 76 KIAS. Locate suitable field. Establish spiral pattern. 1000 ft. above field at downwind position for normal landing approach. When field can easily be reached, slow to 66 KIAS for shortest landing.
Touchdowns should normally be made at lowest possible airspeed with full flaps.
When committed to landing:
Flaps As desired Throttle Close Mixture idle cut-off Magnetos OFF Battery Master switch OFF ALTR Switch OFF Fuel selector OFF
Seat belt and harnesstight

EMERGENCY PROCEDURES	PA-28-181, ARCHER III
FIRE IN FLIGHT	
Source of fire	check
Electrical fire (smoke in cabin):	
Batt. Master switch	OFF
ALTR switch	OFF
Vents	open
Cabin heat	ÔFF
Land as soon as possible.	
Engine fire:	
Fuel selector	OFF
Throttle	
Mixture	idle cut-off
Electric fuel pump	check OFF
Heater and defroster	OFF
Proceed with power off landing procedure.	
NOTE:	
The possibility of an engine fire in	flight is extremely
remote. The procedure given is ger	neral and Pilot judgment
should be the determining factor for	r action in such an
emergency.	
LOSS OF OIL PRESSURE	
Land as soon as possible and investigate cau	se.
Prepare for power off landing.	
LOSS OF FUEL PRESSURE	
T1	ON .
Electric fuel pump	ON

#### HIGH OIL TEMPERATURE

Land at nearest airport and investigate the problem.

Prepare for power off landing.

#### ELECTRICAL FAILURES

#### NOTE:

Anytime the bus voltage is below 25 Vdc, the Low Bus Voltage Annunciator will be illuminated.

ALT annunciator light illuminated:

If ammeter shows zero:

ALT switch.....OFF

Reduce electrical loads to minimum:

ALT circuit breaker......Check and reset

ALT switch as required
ON

If power not restored:

ALT switch.....OFF

If alternator output cannot be restored, reduce electrical loads and land as soon as practical. Anticipate complete electrical failure. Duration of battery power will be dependent on electrical load and battery condition prior to failure.

#### NOTE:

Low Bus Voltage Annunciator will be illuminated.

ELECTRICAL OVERLOAD (Alternator over 20 amps above known electrical load)	
ALT switch ON BAT switch OFF	
If alternator loads are reduced:  Electrical load	
Land as soon as practical.	
NOTE	
Due to increased system voltage and radio frequency noise, operation with ALT switch ON and BAT switch OFF should be made only when required by an electrical system failure.	
f alternator loads are not reduced: ALT switchOFF BAT switchAs required	
Land as soon as possible. Anticipate complete electrical failure.	
SPIN RECOVERY	
Rudderfull opposite to direction of rotation	
Control wheelfull forward while neutralizing ailerons	
Throttle idle Rudder neutral (when rotation stops)	
Control wheel	

## OPEN DOOR

If both upper and side latches are open, the door will trail slightly open and airspeeds will be reduced slightly.

To close the door in flight: Slow airplane to 87 KIAS.
Cabin ventsclose
Storm windowoper
If upper latch is openlatch
If side latch is openpull on armrest while
moving latch handle
to latched position
If both latches are openlatch side latch
then top latel
CARBURETOR ICING
Carburetor heatON
Mixtureadjust for maximum
smoothness
ENGINE ROUGHNESS
Carburetor heat ON
If roughness continues after one min:
Carburetor heatOFF
Mixtureadjust for maximum
smoothness
Electric fuel pumpON
Fuel selectorswitch tanks
Engine gauges
Magneto switches
If operation is satisfactory on either one, continue on that magneto at reduced
power and full RICH mixture to first airport.

Prepare for power off landing.

THIS PAGE INTENTIONALLY LEFT BLANK

### 3.7 AMPLIFIED EMERGENCY PROCEDURES (GENERAL)

The following paragraphs are presented to supply additional information for the purpose of providing the pilot with a more complete under-standing of the recommended course of action and probable cause of an emergency situation.

#### 3.9 ENGINE FIRE DURING START

Engine fires during start are usually the result of overpriming. The first attempt to extinguish the fire is to try to start the engine and draw the excess fuel back into the induction system.

If a fire is present before the engine has started, move the mixture control to idle cut-off, open the throttle and crank the engine. This is an attempt to draw the fire back into the engine.

If the engine has started, continue operating to try to pull the fire into the engine.

In either case (above), if fire continues more than a few seconds, the fire should be extinguished by the best available external means.

The fuel selector valves should be OFF and the mixture at idle cut-off if an external fire extinguishing method is to be used.

#### 3.11 ENGINE POWER LOSS DURING TAKEOFF

The proper action to be taken if loss of power occurs during takeoff will depend on the circumstances of the particular situation.

If sufficient runway remains to complete a normal landing, land straight ahead.

If insufficient runway remains, maintain a safe airspeed and make only a shallow turn if necessary to avoid obstructions. Use of flaps depends on the circumstances. Normally, flaps should be fully extended for touchdown.

If sufficient altitude has been gained to attempt a restart, maintain a safe airspeed and switch the fuel selector to another tank containing fuel. Check the electric fuel pump to insure that it is ON and that the mixture is RICH. The carburetor heat should be ON.

If engine failure was caused by fuel exhaustion, power will not be regained after switching fuel tanks until the empty fuel lines are filled. This may require up to ten seconds.

If power is not regained, proceed with the Power Off Landing procedure (refer to the emergency check list and Paragraph 3.15).

#### 3.13 ENGINE POWER LOSS IN FLIGHT

Complete engine power loss is usually caused by fuel flow interruption and power will be restored shortly after fuel flow is restored. If power loss occurs at a low altitude, the first step is to prepare for an emergency landing (refer to Paragraph 3.15). An airspeed of at least 76 KIAS should be maintained.

If altitude permits, switch the fuel selector to another tank containing fuel and turn the electric fuel pump ON. Move the mixture control to RICH and the carburetor heat to ON. Check the engine gauges for an indication of the cause of the power loss. If no fuel pressure is indicated, check the tank selector position to be sure it is on a tank containing fuel.

When power is restored move the carburetor heat to the OFF position and turn OFF the electric fuel pump.

If the preceding steps do not restore power, prepare for an emergency landing.

If time permits, turn the Left and Right magneto switches OFF then ON one at a time. Move the throttle and mixture control levers to different settings. This may restore power if the problem is too rich or too lean a mixture or if there is a partial fuel system restriction. Try other fuel tanks. Water in the fuel could take some time to be used up, and allowing the engine to windmill may restore power. If power loss is due to water, fuel pressure indications will be normal.

REPORT: VB-1611 ISSUED: JULY 12, 1995

If engine failure was caused by fuel exhaustion, power will not be restored after switching fuel tanks until the empty fuel lines are filled. This may require up to ten seconds.

If power is not regained, proceed with the Power Off Landing procedure (refer to the emergency check list and Paragraph 3.15).

#### 3.15 POWER OFF LANDING

If loss of power occurs at altitude, trim the aircraft for best gliding angle 76 KIAS (Air Cond. off) and look for a suitable field. If measures taken to restore power are not effective, and if time permits, check your charts for airports in the immediate vicinity; it may be possible to land at one if you have sufficient altitude. If possible, notify the FAA by radio of your difficulty and intentions. If another pilot or passenger is aboard, let him help.

When you have located a suitable field, establish a spiral pattern around this field. Try to be at 1000 feet above the field at the downwind position, to make a normal landing approach. When the field can easily be reached, slow to 66 KIAS with flaps down for the shortest landing. Excess altitude may be lost by widening your pattern, using flaps or slipping, or a combination of these.

Touchdown should normally be made at the lowest possible airspeed.

When committed to a landing, lower the flaps as desired, close the throttle, move the mixture to idle cut-off, and shut OFF the magnetos. Turn the battery master and alternator switches OFF. Move the fuel selector valve to OFF. The seat belts and shoulder harness should be tightened.

#### 3.17 FIRE IN FLIGHT

The presence of fire is noted through smoke, smell and heat in the cabin. It is essential that the source of the fire be promptly identified through instrument readings, character of the smoke, or other indications since the action to be taken differs somewhat in each case.

Check for the source of the fire first.

ISSUED: JULY 12, 1995

If an electrical fire is indicated (smoke in the cabin), the battery master switch should be turned OFF. The cabin vents should be opened and the cabin heat turned OFF. A landing should be made as soon as possible.

If an engine fire is present, switch the fuel selector to OFF and close the throttle. The mixture should be at idle cut-off. Turn the electric fuel pump OFF. In all cases, the heater and defroster should be OFF. If radio com-munication is not required, select battery master and alternator switchs OFF. Proceed with power off landing procedure.

#### NOTE

The possibility of an engine fire in flight is extremely remote. The procedure given is general and pilot judgment should be the determining factor for action in such an emergency.

#### 3.19 LOSS OF OIL PRESSURE

Loss of oil pressure may be either partial or complete. A partial loss of oil pressure usually indicates a malfunction in the oil pressure regulating system, and a landing should be made as soon as possible to investigate the cause and prevent engine damage.

A complete loss of oil pressure indication may signify oil exhaustion or may be the result of a faulty gauge. In either case, proceed toward the nearest airport, and be prepared for a forced landing. If the problem is not a pressure gauge malfunction, the engine may stop suddenly. Maintain altitude until such time as a dead stick landing can be accomplished. Don't change power settings unnecessarily, as this may hasten complete power loss.

Depending on the circumstances, it may be advisable to make an off airport landing while power is still available, particularly if other indications of actual oil pressure loss, such as sudden increases in temperatures, or oil smoke, are apparent, and an airport is not close.

If engine stoppage occurs, proceed with Power Off Landing.

REPORT: VB-1611

#### 3.21 LOSS OF FUEL PRESSURE

The most probable cause of loss of fuel pressure is either fuel depletion in the fuel tank selected or failure of the engine driven fuel pump. If loss of fuel pressure occurs, turn ON the electric fuel pump and check that the fuel selector is on a tank containing usable fuel.

If loss of fuel pressure is due to failure of the engine driven fuel pump the electric fuel pump will supply sufficient fuel pressure.

After fuel pressure and power are regained, turn the electric fuel pump OFF. If fuel pressure starts to drop, turn the electric fuel pump ON and land at the nearest suitable airport as soon as possible and have the cause investigated.

#### CAUTION

If normal engine operation and fuel pressure is not immediately re-established, the electric fuel pump should be turned off. The lack of fuel pressure indication could indicate a leak in the fuel system, or fuel exhaustion.

#### 3.23 HIGH OIL TEMPERATURE

An abnormally high oil temperature indication may be caused by a low oil level, an obstruction in the oil cooler, damaged or improper baffle seals, a defective gauge, or other causes. Land as soon as practical at an appropriate airport and have the cause investigated.

A steady, rapid rise in oil temperature is a sign of trouble. Land at the nearest airport and let a mechanic investigate the problem. Watch the oil pressure gauge for an accompanying loss of pressure.

#### 3.25 ELECTRICAL FAILURES

#### NOTE:

Anytime the bus voltage is below 25 Vdc, the Low Bus Voltage Annunciator will be illuminated.

Loss of alternator output is detected through zero reading on the ammeter. Before executing the following procedure, insure that the reading is zero and not merely low by actuating an electrically powered device, such as the landing light. If no increase in the ammeter reading is noted, alternator failure can be assumed.

The electrical load should be reduced as much as possible. Check the alternator circuit breakers for a popped circuit.

The next step is to attempt to reset the overvoltage relay. This is accomplished by moving the ALT switch to OFF for one second and then to ON. If the trouble was caused by a momentary overvoltage condition (30.5 volts and up) this procedure should return the ammeter to a normal reading.

#### NOTE:

Low Bus Voltage Annunciator will be illuminated.

If the ammeter continues to indicate "0" output, or if the alternator will not remain reset, turn off the ALT switch, maintain minimum electrical load and land as soon as practical. Anticipate complete electrical failure. Duration of battery power will be dependent on electrical load and battery condition prior to failure.

# 3.27 ELECTRICAL OVERLOAD (Alternator over 20 amps above known electrical load)

If abnormally high alternator output is observed (more than 20 amps above known electrical load for the operating conditions) it may be caused by a low battery, a battery fault or other abnormal electrical load. If the cause is a low battery, the indication should begin to decrease toward normal within 5 minutes. If the overload condition persists attempt to reduce the load by turning off non-essential equipment.

Turn the BAT switch OFF and the ammeter should decrease. Turn the BAT switch ON and continue to monitor the ammeter. If the alternator output does not decrease within 5 minutes, turn the BAT switch OFF and land as soon as practical. All electrical loads are being supplied by the alternator.

#### NOTE

Due to higher voltage and radio frequency noise, operation with the ALT switch ON and the BAT switch OFF should be made only when required by an electrical failure.

#### 3.29 SPIN RECOVERY

Intentional spins are prohibited in this airplane. If a spin is inadvertently entered, immediately apply full rudder opposite to the direction of rotation. Move the control wheel full forward while neutralizing the ailerons. Move the throttle to IDLE. When the rotation stops, neutralize the rudder and ease back on the control wheel as required to smoothly regain a level flight attitude.

#### 3.31 OPEN DOOR

The cabin door is double latched, so the chances of its springing open in flight at both the top and side are remote. However, should you forget the upper latch, or not fully engage the side latch, the door may spring partially open. This will usually happen at takeoff or soon afterward. A partially open door will not affect normal flight characteristics, and a normal landing can be made with the door open.

If both upper and side latches are open, the door will trail slightly open, and airspeed will be reduced slightly.

To close the door in flight, slow the airplane to 87 KIAS, close the cabin vents and open the storm window. If the top latch is open, latch it. If the side latch is open, pull on the armrest while moving the latch handle to the latched position. If both latches are open, close the side latch then the top latch.

ISSUED: JULY 12, 1995 REPORT: VB-1611

#### 3.33 CARBURETOR ICING

Under certain moist atmospheric conditions at temperatures of -5°C to 20°C, it is possible for ice to form in the induction system, even in summer weather. This is due to the high air velocity through the carburetor venturi and the absorption of heat from this air by vaporization of the fuel.

To avoid this, carburetor preheat is provided to replace the heat lost by vaporization. Carburetor heat should be full on when carburetor ice is encountered. Adjust mixture for maximum smoothness.

#### 3.35 ENGINE ROUGHNESS

Engine roughness is usually due to carburetor icing which is indicated by a drop in RPM, and may be accompanied by a slight loss of airspeed or altitude. If too much ice is allowed to accumulate, restoration of full power may not be possible; therefore, prompt action is required.

Turn carburetor heat on (See Note). RPM will decrease slightly and roughness will increase. Wait for a decrease in engine roughness or an increase in RPM, indicating ice removal. If no change in approximately one minute, return the carburetor heat to OFF.

If the engine is still rough, adjust the mixture for maximum smoothness. The engine will run rough if too rich or too lean. The electric fuel pump should be switched to ON and the fuel selector switched to the other tank to see if fuel contamination is the problem. Check the engine gauges for abnormal readings. If any gauge readings are abnormal, proceed accordingly. Select the Left magneto switch OFF then ON and repeat with the Right magneto switch. If operation is satisfactory on either magneto, proceed on that magneto at reduced power, with mixture full RICH, to a landing at the first available airport.

If roughness persists, prepare for a precautionary landing at pilot's discretion.

#### NOTE

Partial carburetor heat may be worse than no heat at all, since it may melt part of the ice, which will refreeze in the intake system. When using carburetor heat, therefore, always use full heat, and when ice is removed return the control to the full cold position.

REPORT: VB-1611 ISSUED: JULY 12, 1995

3.16