TRINIDAD TB-20

1. The engine's rated horsepower at sea level is				
	a.	250		
	b.	200		
	c.	180		
	d.	275		
2.	The usa	ble fuel capacity is US gallons.		
	a.	86.2		
	b.	88.8		
	c.	52		
	d.	75.2		
3.	Maxim	Maximum oil capacity and minimum safe quantity for the IO-540 engine is		
	quarts.			
	a.	8, 6		
	b.	6, 3		
	c.	10, 4		
		12, 9		
4.	During preflight inspection the fuel tanks should be <i>visually</i> checked to insure the			
	-	d amount of fuel for the planned flight is aboard:		
		True		
		False		
5.		When starting a flooded engine, the mixture control should be in what position?		
		Doesn't Matter		
		Full Rich		
		Idle Cut-Off		
		Half Lean		
6.		the magnetos during run-up at RPM. Differential drop should not exceed		
		RPM while max drop on either magneto is RPM.		
		2000/50/50		
		1700/75/125		
		1800/50/150		
		2000/50/175		
7.		ll warning horn is typically activated at:		
		10-20 knots above stall speed		
		Stall speed		
		5-10 knots below stall speed		
	d.	5-10 knots above stall speed		

8.	With flaps and gear retracted, power at idle, and maximum gross weight, the aircraft			
	should	I stall at KIAS with 0° bank and KIAS at 30°.		
	a.	70, 75		
	b.	80, 75		
	c.	65, 75		
	d.	55, 65		
9.	The m	ixture should always be leaned during cruise at 75% power or less		
	a.	True		
	b.	False		
10.	The pr	rocedure for flying in turbulent air is as follows:		
	a.	Maintain fastest speed possible.		
	b.	Slow to below maneuvering speed and maintain altitude.		
	c.	Slow to below maneuvering speed and maintain level pitch attitude.		
	d.	Don't fly in turbulent air.		
11.	Final a	approach speed is KIAS.		
	a.	73 KIAS		
	b.	85 KIAS		
	c.	80 KIAS		
	d.	70 KIAS		
12.	The fu	el selector should be on fuel tank(s) prior to landing.		
	a.	The Left		
	b.	Both		
	c.	The Fullest		
	d.	The Right		
13.	For a 1	normal takeoff, allow the aircraft to accelerate to KIAS, then ease back on		
	the co	ntrol wheel to establish the approximate pitch attitude for climb at KIAS.		
	a.	68/75		
	b.	50/60		
	c.	40/50		
	d.	65/73		
14.	The el	ectric fuel pump should be "on" for the following operations:		
	a.	Engine Starting Procedure		
	b.	Takeoff		
	c.	Landing		
	d.	Switching of fuel tanks in-flight		
	e.	All of the Above		

15. Maxin	num gross weight and maximum baggage weights are and,
respec	ctively.
a.	2550/200
b.	3086/143
c.	2943/150
d.	2325/200
16. The fl	ap positions are for operation and to be used as an entrance step.
a.	10/20/30 and half down
b.	10/15/30 and full down
c.	Take off (10°/Landing (40°), and not
d.	10/20/40 and makes no sense
17. The el	ectrical system includes a volt battery and a volt, amperes
alterna	ator.
a.	24/28/70
b.	12/14/60
c.	28/28/70
d.	12/12/30
18. The N	ever Exceed speed, maximum structural cruising speed are:
a.	187 KIAS 150 KIAS
b.	189 KIAS 130 KIAS
c.	151 KIAS 129 KIAS
d.	190 KIAS 175 KIAS
19. The m	naneuvering speed for a 3086 lb. aircraft is KIAS.
a.	125
b.	129
c.	132
d.	147
20. Gear e	extension and operating speeds are
a.	The same, extension is 120 mph (CAS) and operating speed is 120 mph (CAS)
b.	The same when the aircraft is at or above 2,000 feet.
c.	Different, extended limit is 139 knots (KIAS) and operating speed is 129 knots
	(KIAS)
d.	Equal to the Vne speed; high performance aircraft do not have gear speed
	restrictions.
21. Oil pro	essure limits are PSI minimum and PSI maximum.
a.	12.5 and 29
b.	15.0 and 50
c.	45 and 80
d.	25.0 and 100

- 22. Fuel pressure operating range is ______ PSI. a. 0.1 to 8
 - b. 1.0 to 8
 - c. 1.0 to 6

 - d. 0.1 to 10
- 23. During ground handling, be certain **NOT** to ______ or _____.
 - a. Turn the nose gear beyond its steering radius, tow when the controls are secured
 - b. Tow with ropes unless positioned low on the wheels; have a fully qualified pilot in the seat
 - c. Push on the inboard edges of the propeller blades, back the aircraft without a spotter
 - d. Pull forward by use of a tow bar alone, refuel before setting parking brake.
- 24. What cruise performance for best power is under the following conditions

O.A.T.: 40°F **PWR:** 65%

Pressure ALT: 6000

- a. 140 TAS
- b. 160 TAS
- c. 150 TAS
- d. 130 TAS
- 25. Find Center of Gravity from the following conditions:

	Weight	Arm	Moment
Aircraft	1903.37	38.96	74125.43
Pilot/PAX	350 lbs	45.38	
Rear PAX	200 lbs	80.00	
Full Fuel		42.70	
Baggage	100 lbs	102.54	

- a. 45.1
- b. 41.4
- c. 46.5
- d. 43.5
- 26. Are touch and goes permissible in an Aero Club owned complex aircraft (See

AFMAN34-232)?

- a. True
- b. False